

# STAR-CCM+ Analysis Results

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## ⊛ General purpose Navier-Stokes code

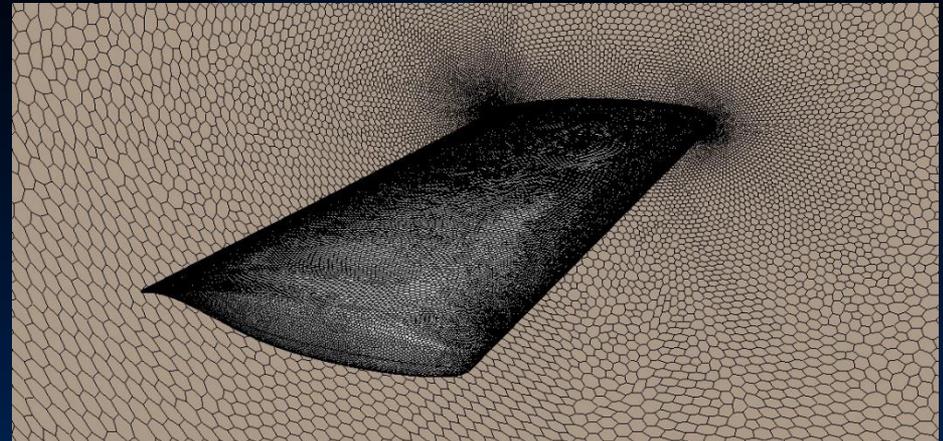
- Cell-centered finite volume

## ⊛ Current analysis

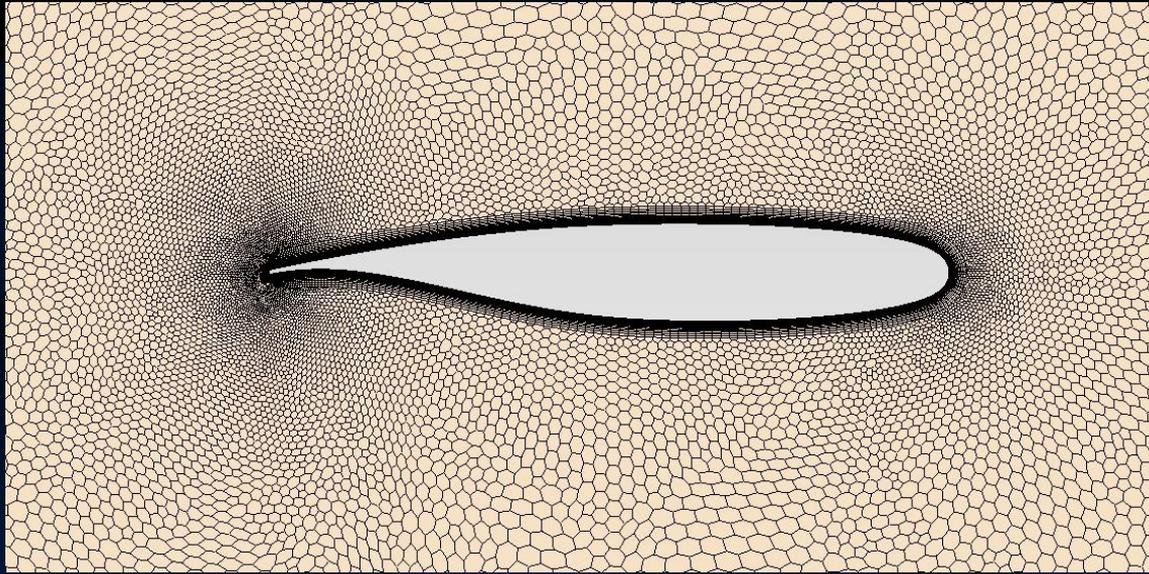
- Coupled solver for mass, momentum, energy
- AUSM+ flux reconstruction
- Venkatakrisshnan slope limiter
- $k-\omega$  SST turbulence model (RANS)
- Pseudo-time-marching for steady flow
- Implicit scheme for unsteady flow

# Geometry and Grid

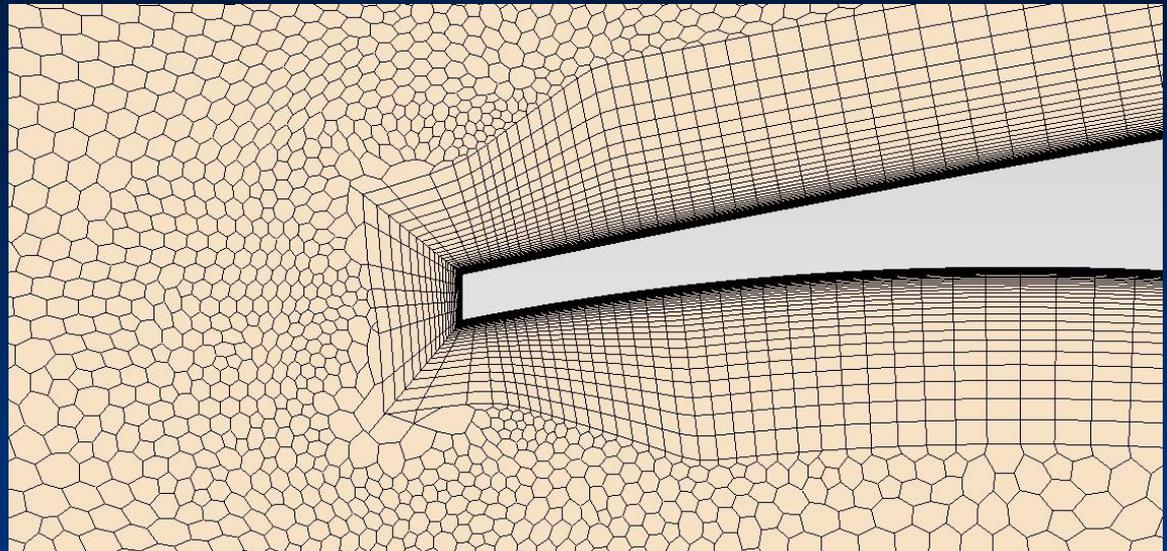
- ⊗ **Hemispherical domain**
  - 3200 in. diameter
- ⊗ **STAR-CCM+ Mesher**
- ⊗ **Arbitrary polyhedral cells**
  - Coarse – 2.49 million cells
  - Medium – 9.10 million
  - Fine – 34.64 million
- ⊗ **Cell near-wall thickness**
  - Coarse –  $\Delta y = 0.000042''$
  - Medium –  $\Delta y = 0.000038''$
  - Fine -  $\Delta y = 0.000032''$
  - Cell  $y^+ \leq 1$  in each case



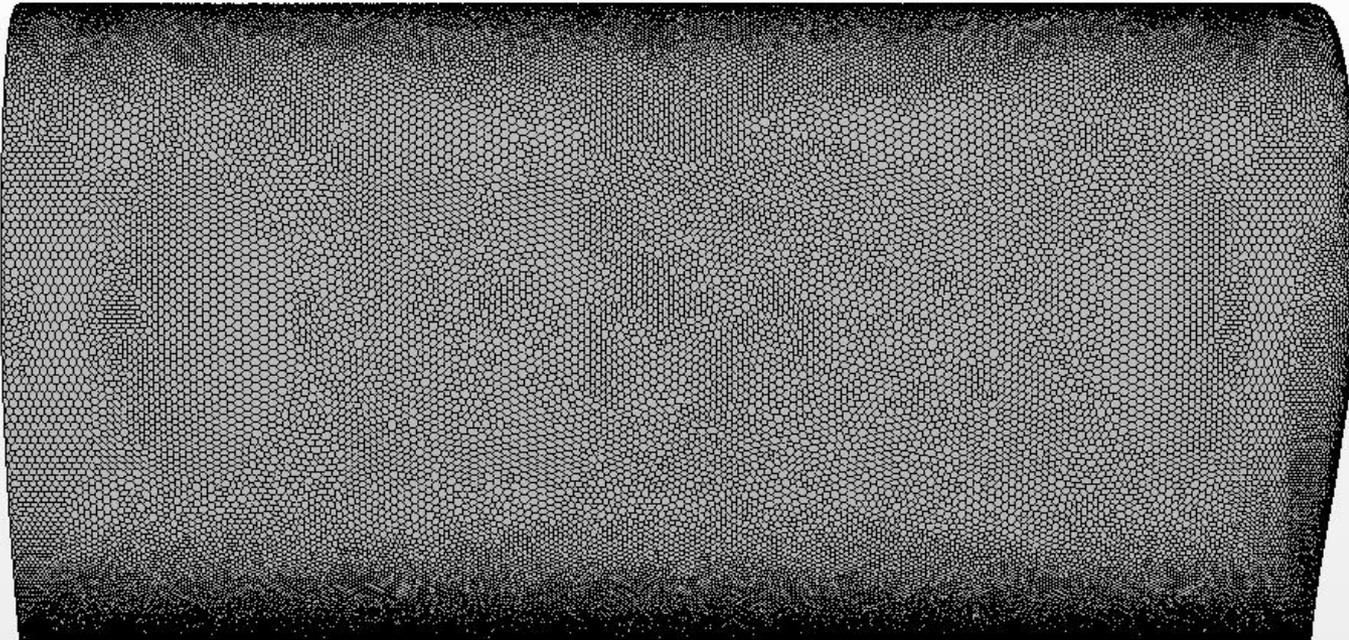
# Mesh – Medium Resolution Example



**Mesh Near Wing**



**Mesh Near Trailing Edge**



## Surface Mesh on Wing

- Mid-chord stream-wise  $\Delta x \approx 0.12''$
- Leading edge stream-wise  $\Delta x \approx 0.05''$
- Trailing edge stream-wise  $\Delta x \approx 0.02''$

## ⊗ Initialization:

- Inviscid flow solution (velocity, pressure, temperature)
- Multigrid solution – 10 levels
- 2 orders of magnitude (relative) drop in residuals

## ⊗ Full Solution:

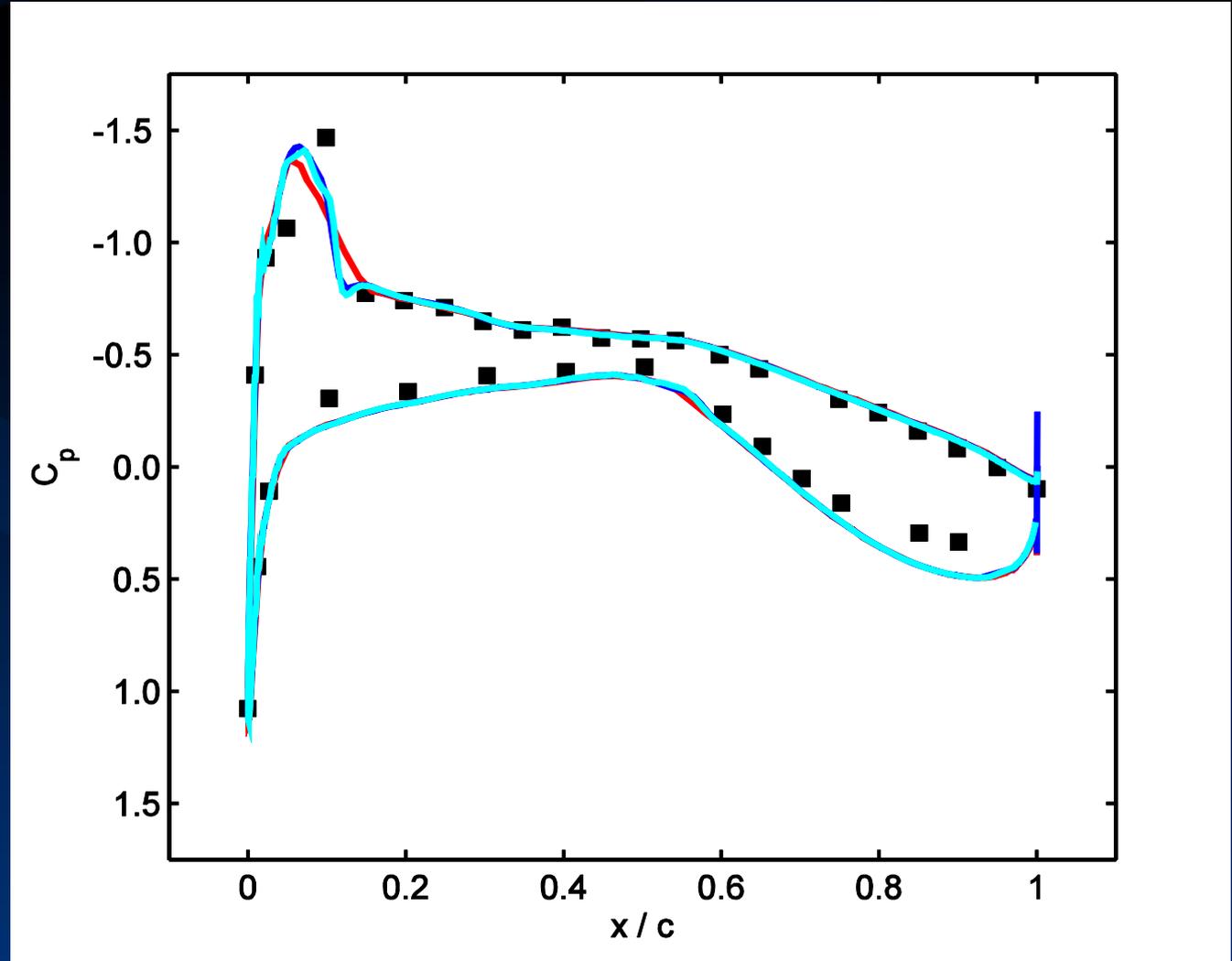
- Additional 5 orders of magnitude (relative) drop in residuals
- Algebraic Multigrid solution of matrix equations

# Case 1 Steady Results



$C_p$  @ 60% Span

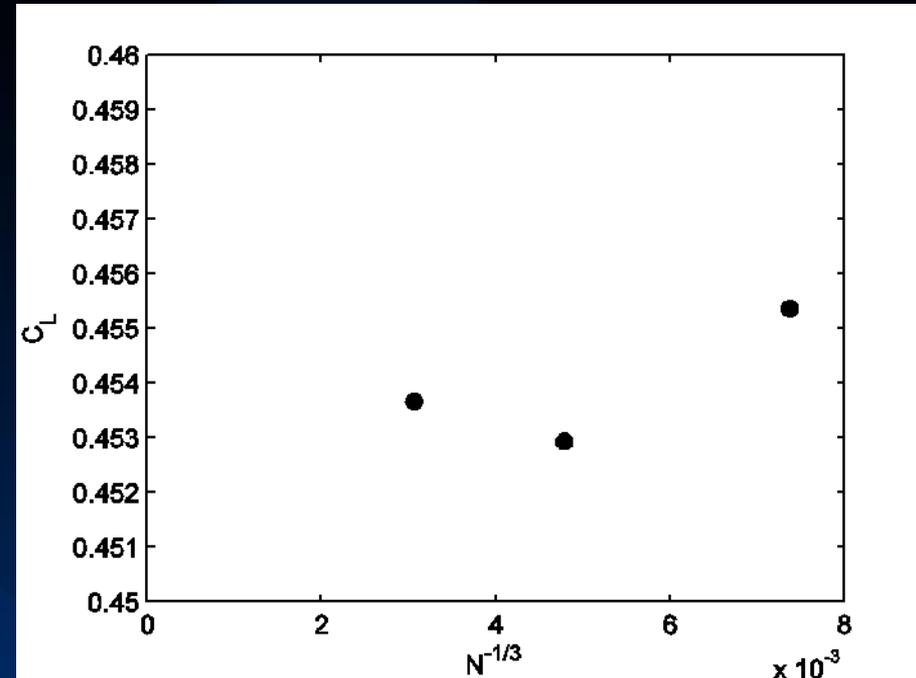
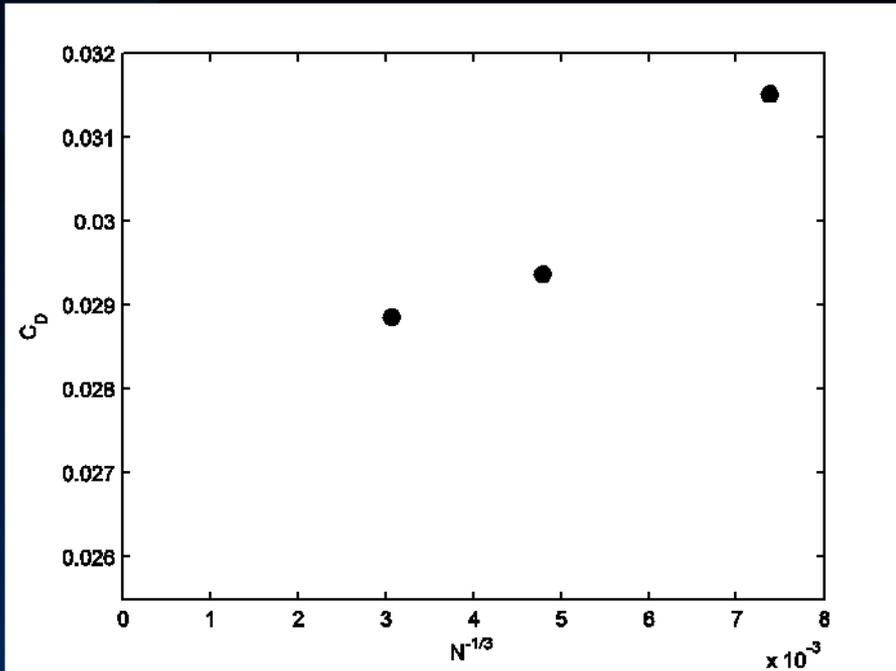
Red – Coarse  
Blue – Medium  
Cyan - Fine



# Case 1 Steady – Mesh Convergence

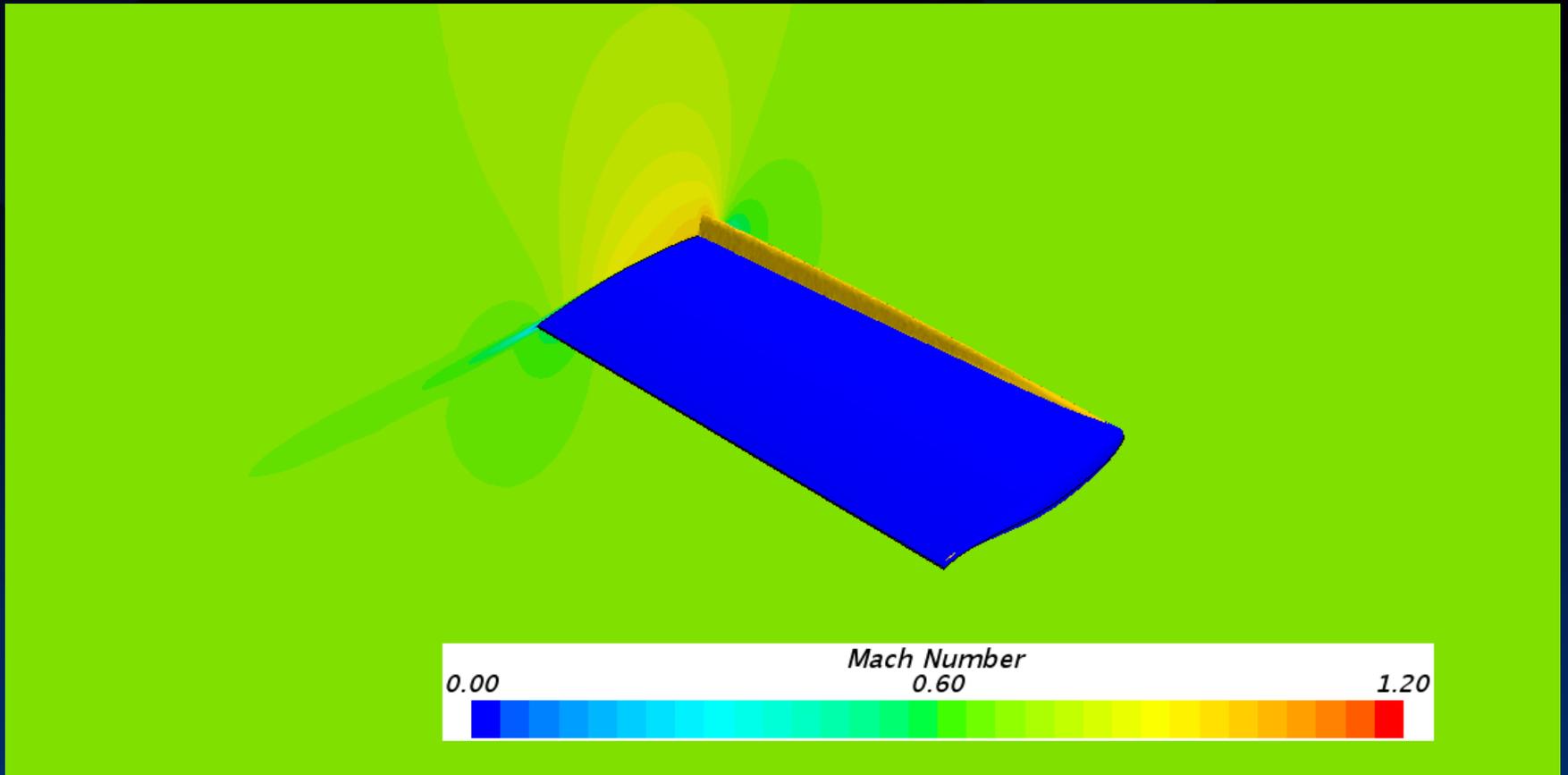


## Total $C_D$ and Total $C_L$ vs. $N^{-1/3}$



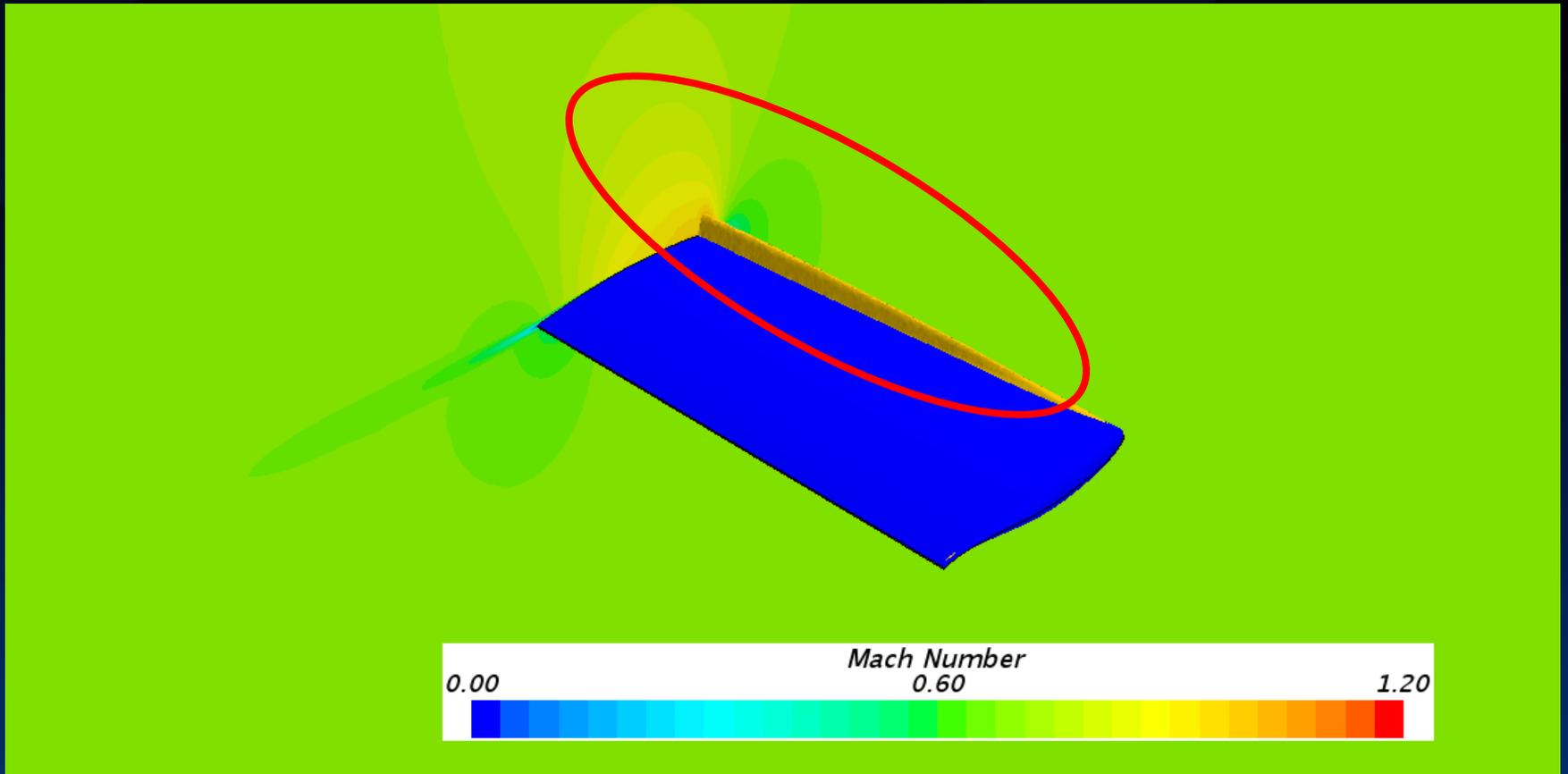
- From Medium to Fine Grid:
  - Less than 2% change  $C_D$
  - Less than 0.5% change  $C_L$

# Case 1 Steady - Visualization



Mach No. @ Medium Resolution Mesh

# Case 1 Steady - Visualization



Mach No. @ Medium Resolution Mesh

# Solution Strategy – Case 1 Unsteady



⊗ **Medium Mesh Spatial Resolution**

⊗ **Refinement in  $\Delta t$**

- 1.0 ms ( 100 steps per period )
- 0.5 ms ( 200 steps per period )
- 0.25 ms ( 400 steps per period )

⊗ **15 Sub-iterations per global time step**

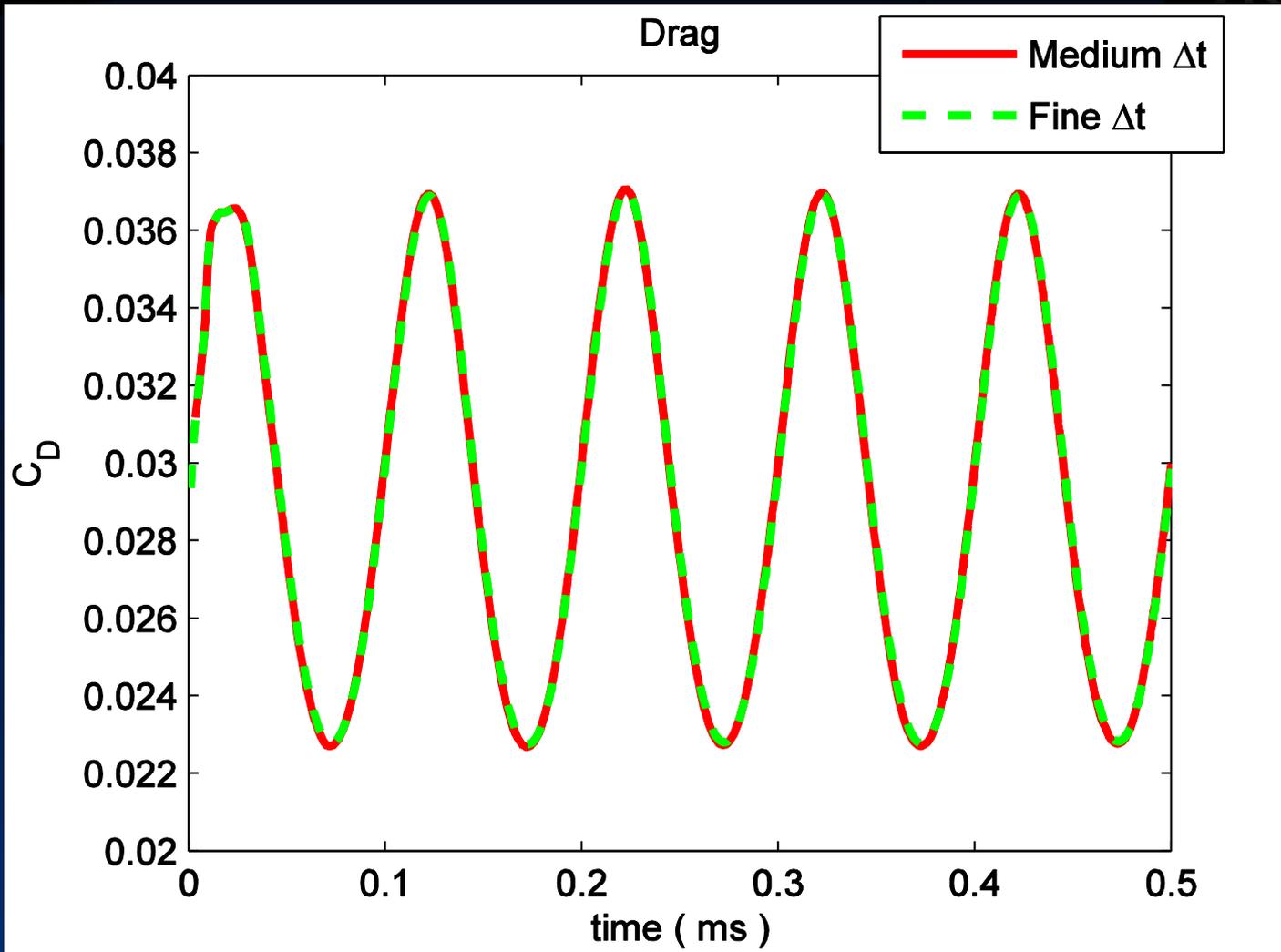
- About 6 orders of magnitude drop in residuals per step

⊗ **2<sup>nd</sup> Order BDF time integration**

⊗ **Specified Rigid Body Rotation for wing**

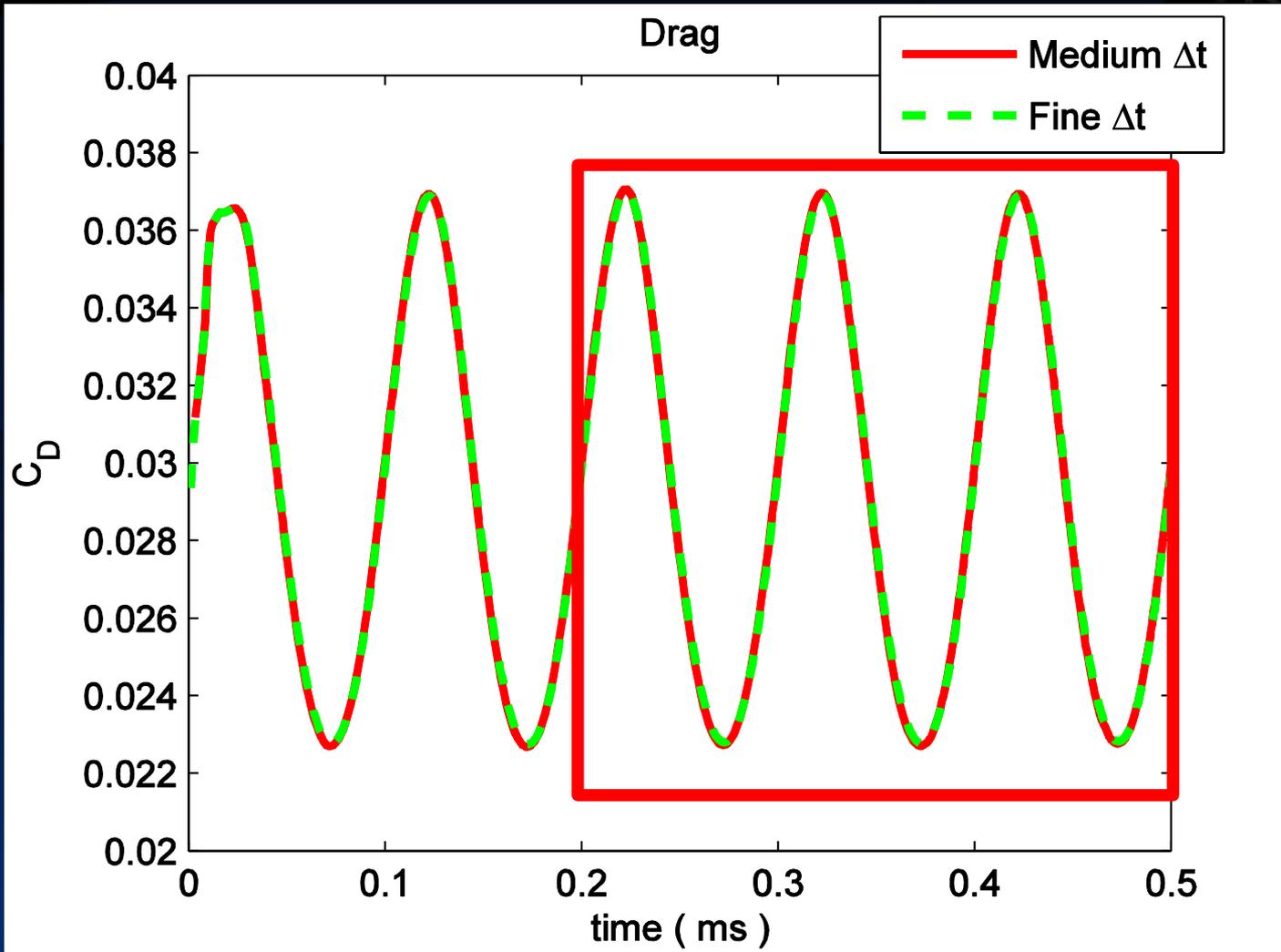
- No Mesh Deformation

# Temporal Convergence Results



Total  $C_D$  vs. time

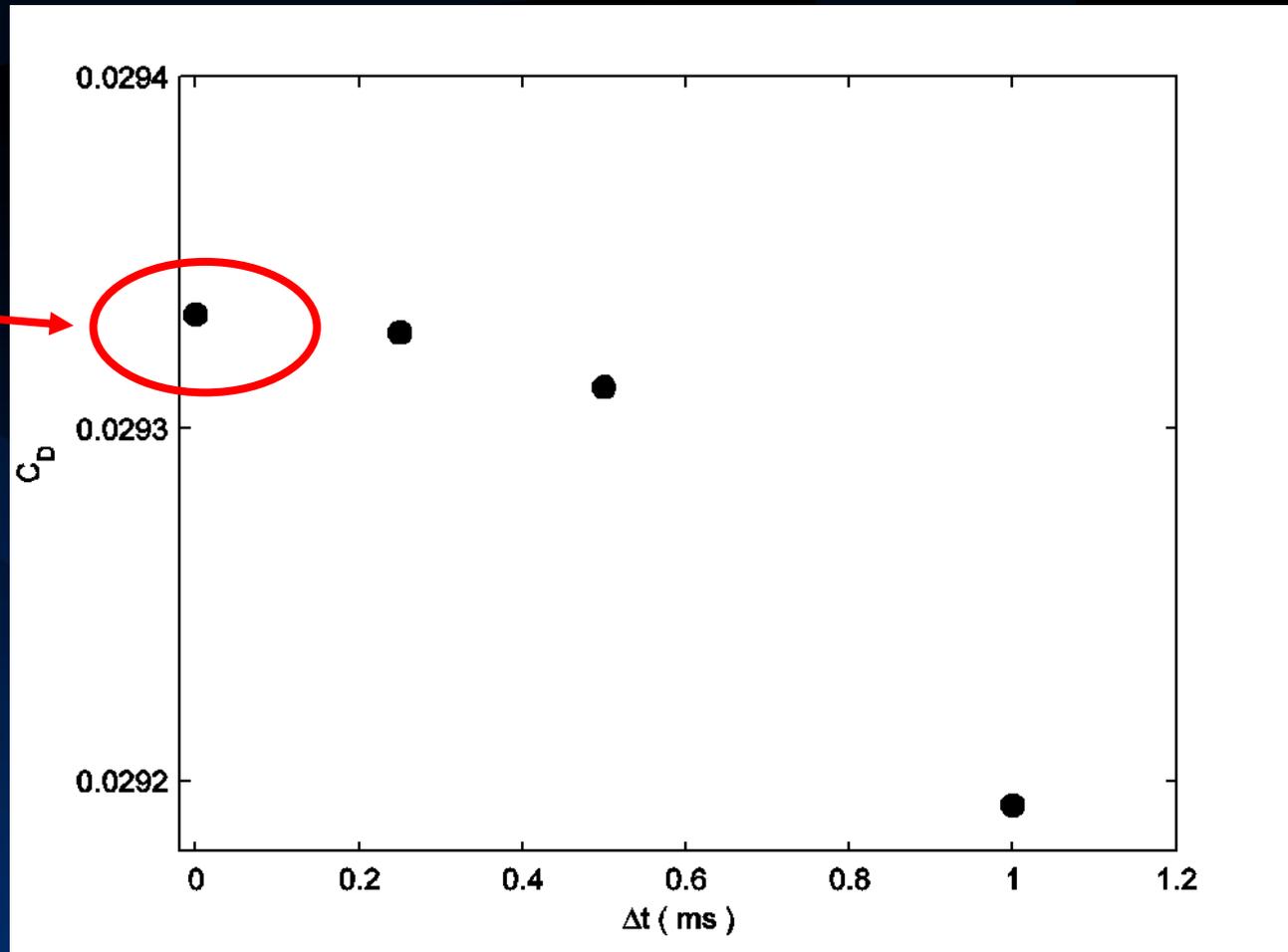
# Temporal Convergence Results



Total  $C_D$  vs. time

# Temporal Convergence Results

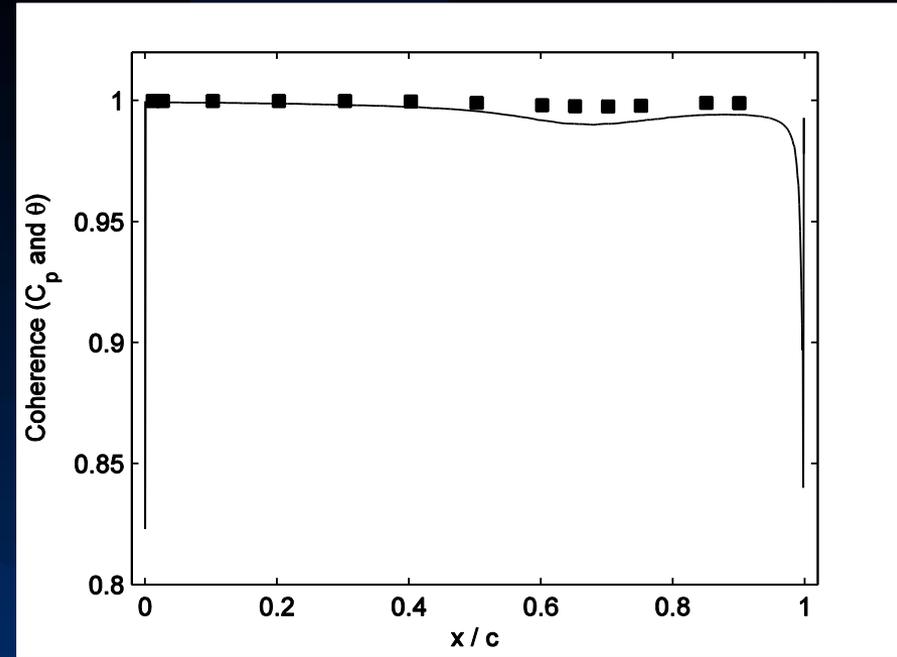
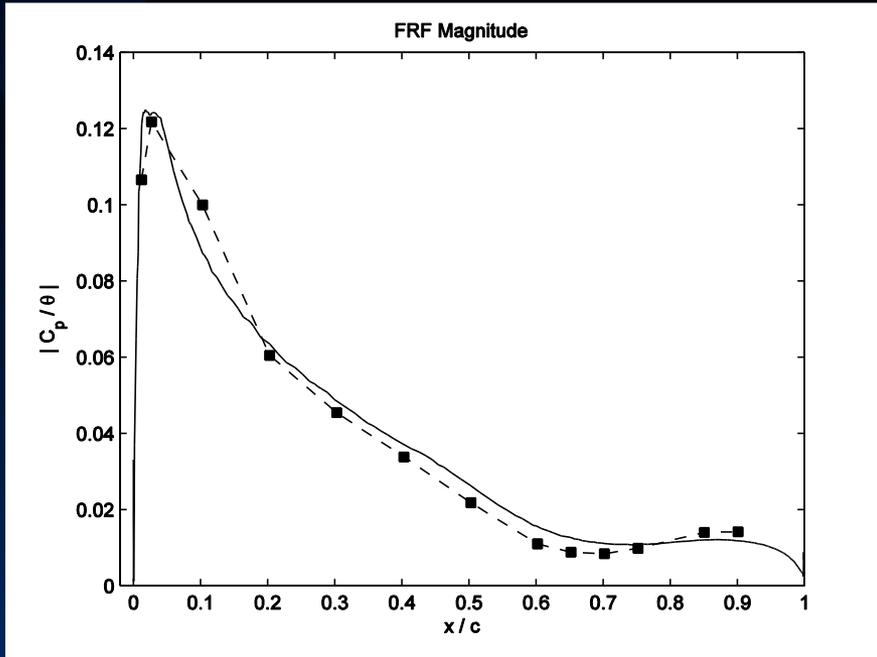
Richardson  
Extrapolation



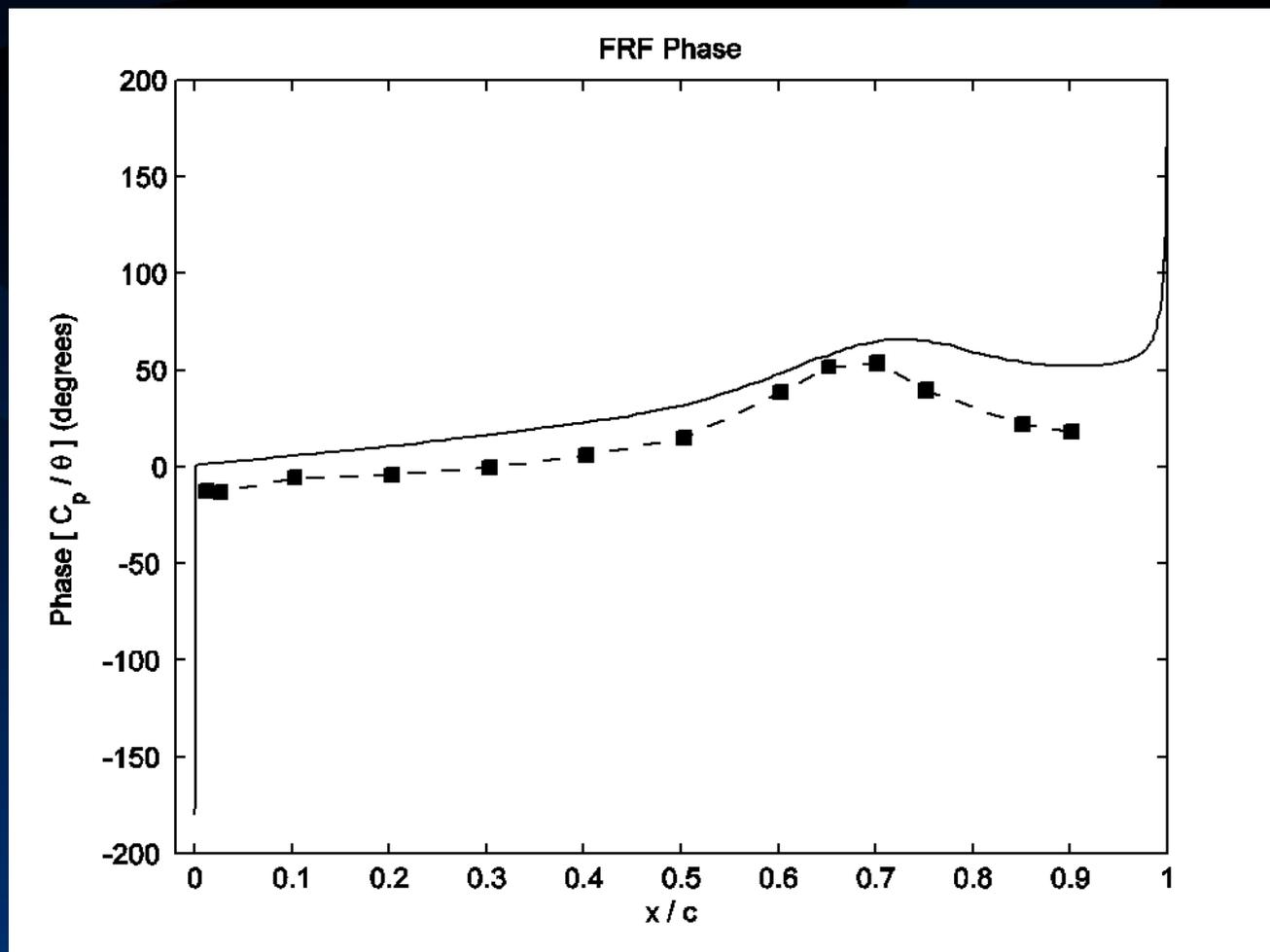
Time-Averaged Total  $C_D$  vs. time step size

$C_D$  within 0.02% of asymptotic value

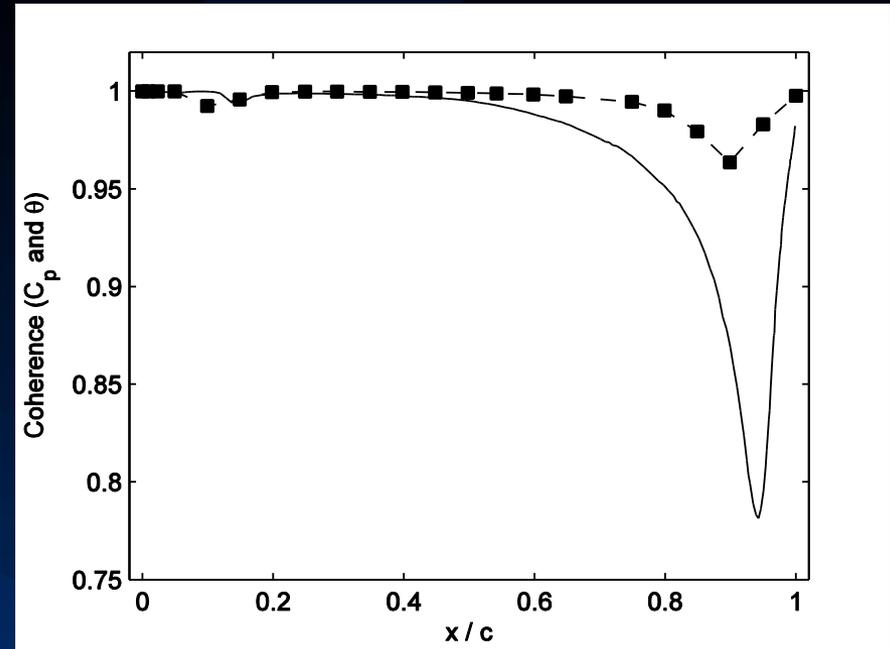
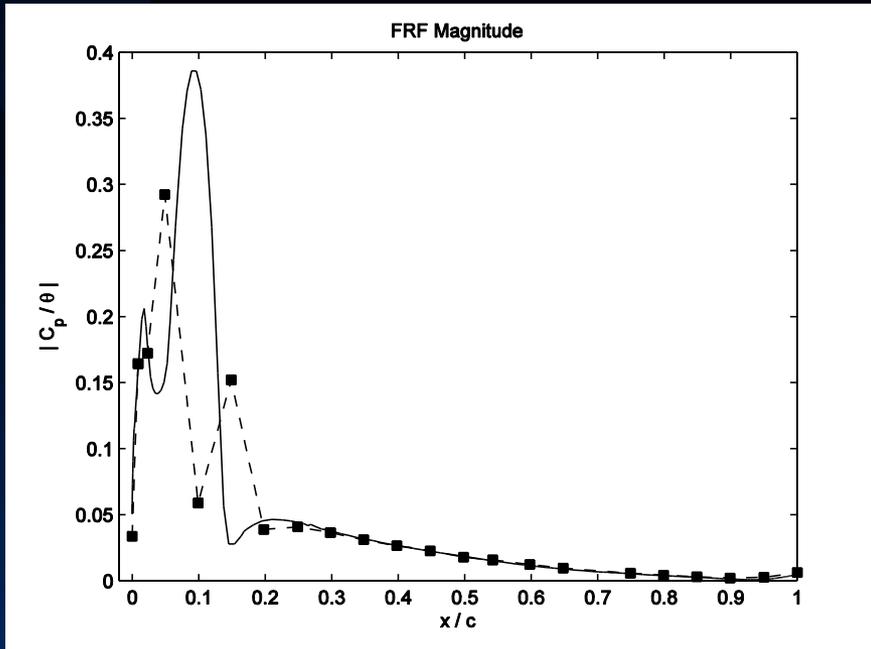
# FRF Results – Lower Surface @ 60%Span



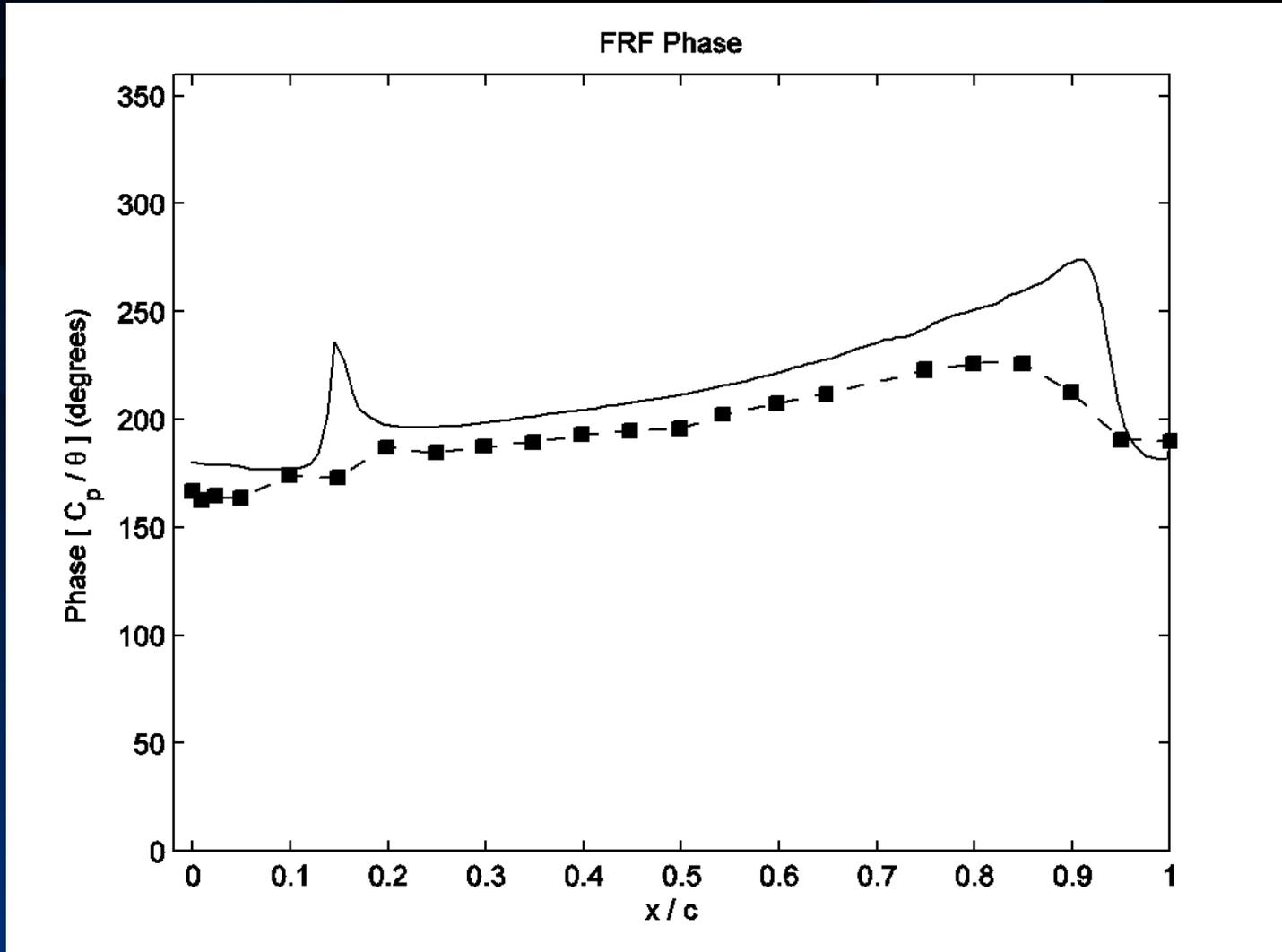
# FRF Results – Lower Surface @ 60%Span



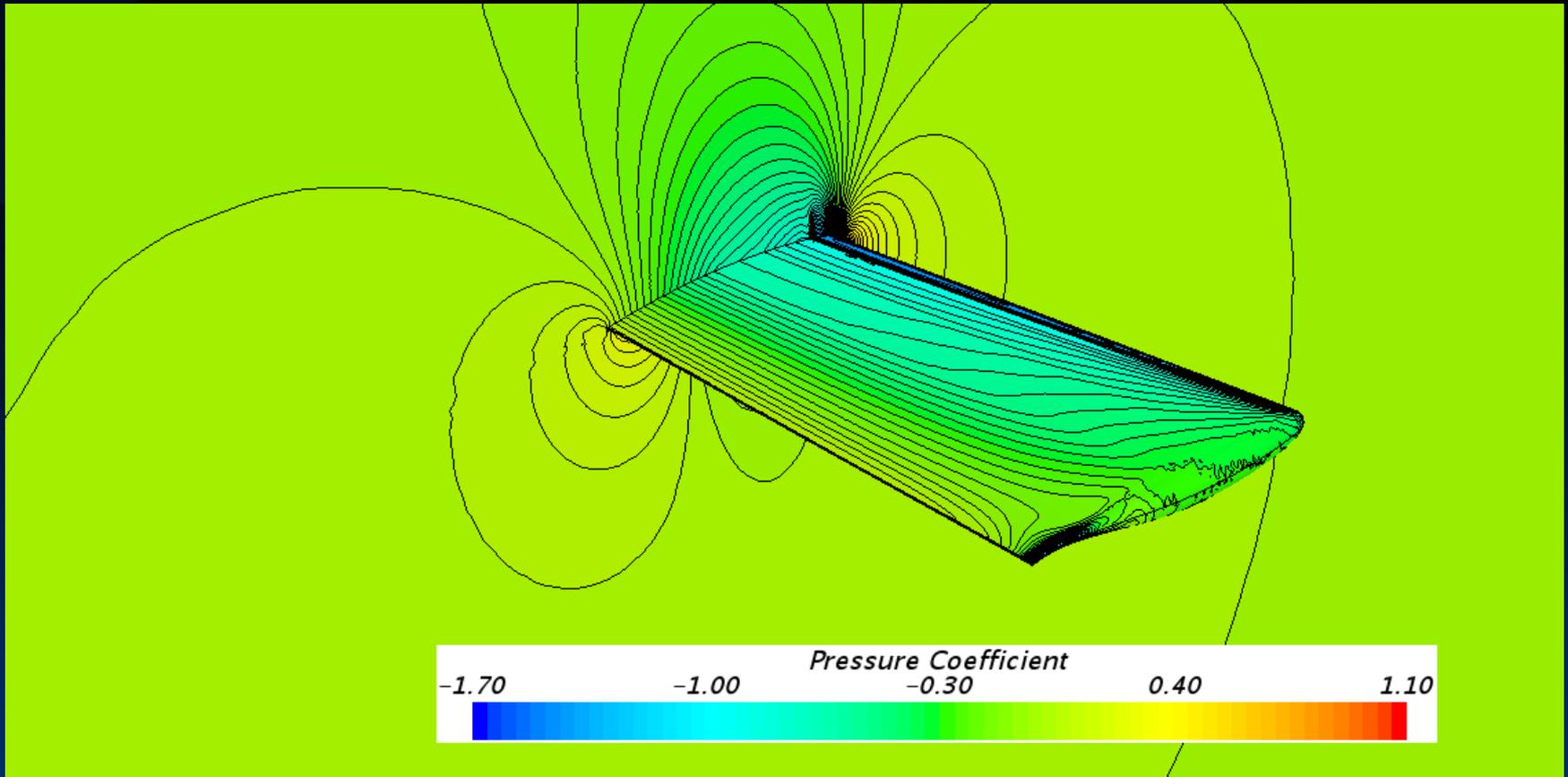
# FRF Results – Upper Surface @ 60%Span



# FRF Results – Upper Surface @ 60%Span



# Case 1 Steady - Visualization



Pressure Coef. @ Medium Resolution Mesh